# FATIGUE EQUATION PARAMETERS RELATIONSHIPS

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### ABSTRACT

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Substantial amounts of data concerning the fatigue performance tests are presented. These data contain the information about results that are expressed by means of the fatigue equation parameters. Different fatigue tests are included using different types of mixes. The collected information was not similar in all aspects. Some studies included variables, which were not considered by others. The different representations of the fatigue equation makes it necessary to find the relations between the parameters of these different equations in order to study all the data together. The collected data were examined statistically to check its consistency; that is to say, they belong to the same population. The Kolmogorov-Smirnov test for goodness of fit was adopted for this purpose.

### الخلاصة

تم تجميع كمية كبيرة من البيانات الخاصة بفحوص اداء الكلل. هذه البيانات تتكون من نتائج الفحوص المختلفة وتمثل عادة بمعاملات معادلة الكلل. تخص البيانات مجموعة مختلفة من فحوص الكلل والتسي تستم باستخدام خلطات اسفلتية مختلفة كما تركز هذه الفحوص على تأثير عوامل مختلفة وحسب الغاية من كل فحص. ونتيجة لذلك اصبح من الواجب ايجاد العلاقات اللازمة لتوحيد المعاملات لمعادلات الكلل المختلفة وذلك لدر اسة العلاقة التي تحكم هذه البيانات. تم قحص البيانات احصائيا لثبوت انتماء كافة المعاملات السي نفس المجتمع وذلك باستخدام طريقة كولموجوروف- سميرنوف لتحليل البيانات.

### **KEY WORDS**

Fatigue, flexible pavement, pavement performance

#### INTRODUCTION

According to the previous studies, there exist a variety of fatigue equations, which were used for the prediction of fatigue performance of flexible pavements. For the controlled strain mode of loading test, the general fatigue equation is (Pell S. F. 1973):

 $N_f = K (f)^{-A}$ 

(1)

Where:

Nf: the number of repetitions of loading till failure at a particular level of initial strain. : the strain that is repeatedly applied on the specimen. K. A: material coefficients.

For the controlled stre..s mode of loading test, the equation given below is used:

FATIGUE EQUATION PARAMETERS RELATIONSHIPS

 $N_f = k (1)^{-a}$ 

Where:

Nf: the number of repetitions of loading till failure.

Q: the stress that is repeatedly applied on the specimen

k, a : the material coefficients.

The importance of the tensile strain in the fatigue cracking makes it necessary to express both the strain and stress controlled tests in the log strain against log number of repetitions to failure [Equation (1)]. Accordingly, the first equation was adopted by most fatigue studies for both controlled strain and stress modes of loading. Noting that the strain ( value for the case of the controlled stress is the initial strain due to the repeatedly applied stress on the specimen.

It is worthwhile mentioning that the fatigue equations were also found in literature in an inverse presentation (strain as a function of the number of load application) as follows:

 $U = K1 (N_f)^{-A1}$ 

Furthermore, they are found similar to Equation (1) but the strain value is measured in microns as follows:

 $N_f = K2 (\Box \Box)^{-A2}$ 

### Fatigue equation parameters

The constants for all the mentioned above equations depend on the material properties. Since the first equation is widely adopted in fatigue studies, it will be adopted for the study. Thus, the constants for the Equations (3) and (4) are related to those of Equation (1) in order to use all the available fatigue performance data in one form. The relations between constants are obtained as follows:

From Equation (1):	
$Log (N_f) = Log (K) - A Log (\Box)$	(5)
From Equation (3):	
$Log(1) = Log(K1) - A1 Log(N_f)$	(6)
$Log(N_1) = (1/\Lambda 1) Log(K1) - (1/\Lambda 1) Log()$	(7)
Comparing Equation (5) with (7) indicates that:	
A = 1/A1	(8)
In a similar way, the value of (K) can be related to (K1) as shown below:	
Log(K) = (1/A1) Log(K1)	(9)
Log (K) = (1/A1) Log (K1) $Log (K) = Log (K1)^{1/A1}$	(10)
$K = K1^{1/A1}$ or	(11)
$K = K1^{A}$	(12)

The variable reduction procedure described as above is used to find the relations between (A) and (A2) and (K) and (K2):

From Equation (4):	
$Log(N_f) = Log(K2) - A2 Log(DD)$	(13)
$Log(N_f) = Log(K2) - A2 (Log(10^6) + Log(1))$	(14)
$Log(N_f) = Log(K2) - Log(10^{6A2}) - A_2 - Log(1)$	(15)
$Log(N_f) = Log(K2/10^{6A2}) - A2 Log()$	(16)

(4)

(3)

(2)

0	Number 3	Volume 11	September 2005	Journal of Eng	gineering
$Log(N_f) = 1$	Log (K2 *10 <sup>-6A)</sup> Equation (5) w	<sup>2</sup> ) - A2 Log ( ]	) 	(17)	
					18) 19)
Log(K) = 1 K = K2 * 10	$\log (K_2 + 10 + 2)$ $Q^{-6A2} = Or$				20)
$K = K_2 * 1$	0.04			. (.	21)

Depending on the presented as above conversions, all the collected data are presented in a unified mathematical form as given by Equation (1) in order to use the parameters of the equation as the dependent variables in any regression analysis.

### Collection of data

The available studies for the fatigue performance of flexible pavements were taken under consideration. All the fatigue line equation parameters and test circumferences are considered input for fatigue analysis. The collected data consisted of many types of asphalt mixes including regular asphalt concrete (as the major part of the date), hot rolled asphalt, dense graded macadam and many others. Many types of fatigue tests, depending on the type of loading, are included within the data like normal bending, axial bending and wheel loading. The adopted data for the fatigue performance analysis, presented in **Table (1)**, include the fatigue equation parameters from the different types mentioned earlier. These parameters of the adopted fatigue equation are calculated when other forms than the regular fatigue equation (N=K\*)<sup>-A</sup>) is found depending on the conversion equations that was obtained previously.

### The Kolmogorov-Smirnov test for data

In order to examine the similarity between the collected data, the Kolmogorov-Smirnov (K-S) test (Lindgren 1968, Gibbons 1971, Willemsen 1974 and Harnett 1975) is adopted which involved a comparison between cumulative relative frequency distributions of two samples. To make this comparison, the data is put into classes, which have been arrayed from the lowest class to the highest class. If the cumulative relative frequency for each category of sample (1) data distribution is ( $F_i$ ) and ( $f_i$ ) to denote the comparable value for the sample (2) frequency, then the (K-S) test is based on the maximum value of the absolute deference between ( $F_i$ ) and ( $f_i$ ). Accordingly:

(22)

### $D = Max_i (|F_i - f_i|)$

Where:

D : the statistic of the Kolmogorov-Smirnov test.

Fi : the cumulative relative frequency of the ith class in sample (1)

 $f_i$ : the cumulative relative frequency of the ith class in sample (2)

The decision to reject the null hypotheses  $H_0$  (that the samples come from the same population) is based on the value of (D). The larger (D) is, the more confidence we have that  $H_0$  is false. The general procedure for the (K-S) test is to calculate the (D) values for each class and these values are compared with the appropriate critical values of (D). The critical values of (D) are obtained from the available tables in many statistical resources (Gibbons 1971 and Harnett 1975). If the maximum value of (D) is less than the critical value, then  $H_0$  is not rejected. In other words, the agreement between the data of the two samples is sufficiently close enough to believe that they come from the same distribution. It is worthwhile mentioning that the (K-S) test was adopted instead of the Chi-Square test because of many reasons.

The class width and number of classes have an effect on chi-square statistics (Gibbons 1971). Furthermore, (K-S) test can be applied for any sample size, while the chi-square statistic should be

used only for large sample size and each expected cell frequency not too small. Figures 1 and 2 presents graphical presentation for the cumulative relative frequencies of the examined data and **Table (2)** shows the Kolmogorov-Smirnov test results. These results indicate that, the collected data are sufficiently close enough to believe that they come from the same distribution.

### CONCLUSIONS

It is essential to convert the fatigue equations in the collected data to one type before the analysis process since many forms of fatigue equations are used for the presentation of the fatigue performance of flexible pavements. Although the required data for analysis are collected from different fatigue tests using different types of mixes, the results of the statistical test shows that the collected data are sufficiently close enough to believe that they come from the same distribution.

### REFERENCES

Bonnaure, F., Gravois, A. and Udron, J. (1980), A New Method for Predicting the Fatigue Life of Bituminous Mixes, Proceeding of Association of Asphalt Paving Technologists (AAPT), Volume 49, pp 499-528.

Bonnaure, F.P., Huibers, A.H.J.J. and Boonders, A. (1982), A Laboratory Investigation on the Fatigue Characteristics of Bituminous Mixes, Proceeding of Association of Asphalt Paving Technologists (AAPT), Volume 51, pp 104-126.

Epps, J.A. and Monismith, C.L. (1971), Fatigue of Asphalt Concrete Mixes-Summary of Existing Information. American Society of testing and Materials (ASTM), Special Publication 508, Fatigue of Compacted Bituminous Aggregate Mixtures, ASTM, Philadelphia, pp 19-45.

Gibbons, J.D. (1971), Nonparametric Statistical Inference, McGraw- Hill Book Company, New York, U.S.A.

Harnett, D.L. (1975), Introduction to Statistical Methods, Second Edition, Addison-Wesley Publishing Company, Massachusetts, U.S.A.

Hsu, T., and Tseng, K. (1996), Effect of Rest Periods on Fatigue Response of Asphalt Concrete Mixtures, Journal of Transportation Engineering, Volume 122, No. 4, pp 316-322.

Irwin, L.H. (1977), Use of Fracture Energy As a Fatigue Failure Criterion, Proceeding of Association of Asphalt Paving Technologists (AAPT), Volume 46, pp 41-63.

Kallas, B.F. and Puzinauskas, V.P. (1971). Flexure Fatigue Tests of Asphalt Paving Mixtures, American Society of testing and Materials (ASTM), Special Publication 508, Fatigue of Compacted Bituminous Aggregate Mixtures, ASTM, Philadelphia, pp 47-65.

Lindgren B.W. (1968), Statistical Theory, Second Edition, Macmillan Publishing Co., Inc., New York, U.S.A.

Pell, P.S. and Cooper, K.E. (1975), The Effect of Testing and Mix Variables on the Fatigue Performance of Bituminous Paving Materials, Proceeding of Association of Asphalt Paving Technologists (AAPT), Volume 44, pp1-37.

Ref.

Pell, S.F. (1973), Characteristics of Fatigue Behavior, Highway Research Board, Special Report 140, Proceeding of a Symposium on Structural Design of Asphalt Concrete Pavement to Prevent Fatigue Cracking, pp.49-64.

Read, J.M. and Collop A.C. (1997), Practical Fatigue Characterization of Bituminous Paving Mixtures, Proceeding of Association of Asphalt Paving Technologists (AAPT), Volume 66, pp 74-108.

Rowe, G.M. and Brown S.F. (1997), Validation of Fatigue Performance of Asphalt Mixtures with Small Scale Wheel Tracking Experiments, Journal of the Association of Asphalt Paving Technologists (AAPT), Proceedings of the Technical Sessions, Volume 66, pp 31-73.

Tayebali, A.A., Deacon, J.A. and Monismith, C.L. (1995), Development and Evaluation of Surrogate Fatigue Models for SHRP A-003A Abridged Mix Design Procedure. Proceeding of Association of Asphalt Paving Technologists (AAPT), Volume 64, pp 340-365.

Van Dijk, W. (1975), Practical Fatigue Characterization of Bituminous Mixes, Proceeding of Association of Asphalt Paving Technologists (AAPT), Volume 44, pp38-74.

Van Dijk, W. and Visser, W. (1977), The Energy Approach to Fatigue for Pavement Design, Proceeding of Association of Asphalt Paving Technologists (AAPT), Volume 46, pp 1-41.

Willemsen, E.W. (1974), Statistical Reasoning, W.H. Freeman and Company, San Francisco, U.S.A.

Yuce, R. and Monismith, C.L. (1974), Prediction of Load Association Cracking in Pavement Slabs from Laboratory Determined Fatigue Data, Proceeding of Association of Asphalt Paving Technologists (AAPT), Volume 43, pp 332-349.

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....... -O- Kallas et. al. 1971 ----- Peil et. al. 1975 ----- CRGC -+- Pell et al 1975 -\*. a ret .... .... -140 . J ---- -----.......... 100 ---- CRGC ---- KLSA ~ \* \* \* \* 210 11111 ....... ---- LCPC ---- CRR \*\* \*\* \*\* \*\* - Bonnau re et. al. 1982 2.00 \*\*\* Mid class Interval .... 1.000 -..... 1.00 Mid class interval .......... .......... ---- Hsu et. al. 1995 ---- Bormaure et. al. 1982 ---- Hau et. al. 1995 Read et. al. 1997 ............ ----- Rowe et. al. 1997 ----- Van Dijk et. al. 1977 ---- Read et. al. 1997 Rowe et al. 1997 1.78 .... \*\*\* --30-6-200 ........... ----- Van Dijk et. al. 1977 ----- Yuce et. al. 1974 Mid class interval

# Figure 1, Comparison of observed cumulative relative frequencies of examined data.

606

Ref.	Mode	• KI	AI	K	A
Epps et al.	C. Stress	1.30E-03	0.19	1.19E-15	5.17
1971	al lander	1.70E-03	0.21	3.19E-14 *	1.01
	and the	3.10E-03	0.26	2.42E-10	3.83
		7.90E-03	0.29	7.43E-08	3.39
Kallas	C. Stress			2.73E-07	3.25
et al. 1971	unuir da			2.01E-05	2.69
				1.37E-06	3.27
	the state of the			1.66E-05	2.72
Contraction of the second				6.52E-05	2.50
111				2.32E-09	3.99
	10-10-53			4.00E-06	3.08
				1.40E-06	3.45
	manin			8.19E-07	3.15
	100 100 500			2.52E-09	3.58
Pell et al.	C. Stress			3.20E-15	5.10
1975	muu			1.30E-13	4.60
1910	10110101			1.30E-10	3.90
		1		2.50E-13	4.60
				4.00E-17	5.50
	avia			1.30E-13	4.50
	1001.100			7.90E-12	4.20
				1.60E-14	4.70
				3.20E-08	3.20
				1.00E-07	3.10
				1.00E-15	5.20
				2.50E-12	4.00
				1.50E-10	3.80
			*****	1.00E-09	3.50
				7.90E-13	4.20
	101107			and the state of t	the state of the second state of the second
	10.01.00		*****	2.00E-10	3.60
	THE OFFICE			1.00E-09	3.60
				5.00E-20	6.30
and and				2.50E-14	4.90
				2.50E-14	4.90
	astus at			1.30E-14	5.10
1.1.1	12120101			1.00E-15	5.50
	====			5.00E-13 .	. 4.70
	matia			5.00E-10	3.40
				1.001:-07	2.70
				3.10109	3.00
	an an an			1.30E-10	3.40
				2.00E-11	3.80
	inter su			5.00E-07	2.50

Table (1) The collected data from previous fatigue performance studies.

Ref .: Data reference. .

Mode: Loading mode (controlled strain or controlled stress).

K, A: Parameters of the general fatigue equation form  $(N=K^*\square^{-A})$ .

K1, A1: Parameters of the fatigue equation form ( $\Box = K1 * N^{-A1}$ ).

### FATIGUE EQUATION PARAMETERS RELATIONSHIPS

Ref.	Mode	K1	A1	K	A.
Pell et al.	C. Stress			1.30E-09	3.20
1975				1.60E-15	4.40
				5.00E-08	2.80
	111 194 km			5.00E-18	5.80
	101.03.93			1.60E-15	5.30
	ur er en			2.50E-17	5.80
	to of its			1.60E-12	4.50
	21/22/2			3.37E-12	4.10
	-25° - 5° 5			4.26E-00	3.40
	+ ++			2.26E-11	3.90
•				1.94E-07	2.90
201			*****	2.38E-05	2.50
Van Dijk	C. Stress			5.75E-05	2.71
1975	352-108 BD			6.92E-05	2.66
	atum (1)			5.49E-04	2.53
Bonnaure	C. Stress	1.24E-03	0.19	5.04E-16	5.26
et al. 1980	011010	1.65E-03	0.20	1.22E-14	5.00
CRGC		1.24E-02	0.32	1.10E-06	3.13
	someter	1.75E-03	0.23	1.03E-12	4.35
		1.14E-03	0.18	4.46E-17	5.56
	dersant	3.09E-03	0.29	2.21E-09	3.45
1.1.1		7.70E-04	0.12	1.13E-26	8.33
	20,222	5.90E-03	0.24	5.15E-10	4.17
	100.000 000	5.60E-04	0.10	3.03E-33	10.00
	300 700 900	1.19E-03	0.13	3.19E-23	7.69
		3.60E-03	0.23	2.37E-11	4.35
	an inter a la l	7.50E-04	0.14	4.78E-23	7.14
		7.40E-04	0.18	4.04E-18	5.56
		5.70E-04	0.11	3.22E-30	9.09
		6.00E-04	0.13	1.65E-25	7.69
	merita and	3.10E-04	0.08	1.39E-44	12.50
CRGC	C. Strain	1.31E-03	0.18	9.66E-17	5.56
		4.75E-03	0.23	7.92E-11	4.35
6	===	8.63E-03	0.23	1.06E-09	4.35
		2.33E-03	0.22	1.08E-12	4.55
	site ingenes	1.38E-03	0.18	1.29E-16	5.56
	100 Mil 401	1.68E-03	0.20	1.34E-14	5.00
	and the last	2.07E-02	0.26	3.33E-07	3.85
	antinina	3.50E-03	0.19	1.19E-13	5.26
	-	6.30E-04	0.12	2.13E-27	8.33
		1.08E-03	0.20	1.47E-15	5.00
		4.40E-04	0.13	1.51E-26	7.69
		and the second state of th	the last had a set of the set of the set		

### Table (1) Continued.

Ref.: Data reference.

Mode: Loading mode (controlled strain or controlled stress).

K, A: Parameters of the general fatigue equation form  $(N=K^*\Box^{-A})$ .

K1, A1: Parameters of the fatigue equation form ( $\Box = K1*N^{-A1}$ ).

CRGC: Grand-Couronne Shell Research Center

32

Table (1) Continued.									
Ref.	Mode	K1	Al	К	А				
Bonnaure	C. Strain	7.40E-04	0.14	4.34E-23	7.14				
et al. 1980		1.38E-03	0.16	1.33E-18	6.25				
CRGC	-	1.04E-03	0.14	4.93E-22	7.14				
		8.00E-04	0.14	7.57E-23	7.14				
		7.90E-04	0.15	2.08E-21	6.67				
01.2	===	1.26E-03	0.18	7.78E-17	5.56				
	an end int	9.50E-04	0.16	1.29E-19	6.25				
	NU 201 DE	6.80E-04	0.12	4.02E-27	8.33				
	12,000 400	9.50E-04	0.12	6.52E-26	8.33				
6.2	mda	7.50E-04	0.14	4.78E-23	7.14				
	-	3.62E-03	0.23	2.43E-11	4.35				
	2.35	1.42E-03	0.19	1.03E-15	5.26				
		1.40E-03	0.17	1.63E-17	5.88				
Bonnatire	C. Strain	3,111-03	0,19	6.37E-14	5.26				
et al. 1980	and some pro-	3.1912-03	0.19	7.28E-14	5.26				
KSLA		3.96E-03	0.18	4.51E-14	5.56				
		3.11E-03	0.14	1.23E-18	7.14				
· · · Z	200.000.000	2.54E-03	0.20	1.06E-13	5.00				
	10.00 m	2.06E-03	0.17	1.58E-16	5.88				
4		7.04E-03	0.19	4.69E-12	5.26				
		2.14E-03	0.18	1.48E-15	5.56				
25.7	====	1.26E-03	0.16	7.54E-19	6.25				
		9.09E-03	0.24	3.12E-09	4.17				
		5.62E-03	0.27	4.63E-09	3.70				
	TEL SAL TEL	3.35E-03	0.20	4.22E-13	5.00				
		4.37E-03	0.20	1.59E-12	5.00				
	an cara	1.73E-03	0.16	5.47E-18	6.25				
		2.85E-03	0.22	2.70E-12	4.55				
		6.61E-03	0.23	3.33E-10	4.35				
		2.09E-03	0.18	1.29E-15	5.56				
		2.47E-03	0.19	1.89E-14	5.26				
	7 22.	3.85E-03.	0,20	8.46E-13	5.00				
•	=	5.02E-03	0.19	7.91E-13	5.26				
		1.43E-03	0.17	1.85E-17	5.88				
	====	3.65E-02	0.36	1.01E-04	2.78				
		2.81E-03	0.19	3.73E-14	5.26				
		9.60E-03	0.24	3.92E-09	4.17				
	====	3.64E-03	0.20	6.39E-13	5.00				
		3.61E-03	0.22	7.90E-12	4.55				
CRR*	C. Stress	1.58E-03	0.20	9.85E-15	5.00				
		2.08E-03	0.22	6.45E-13	4.55				
	are un are	1.10E-03	0.19	2.68E-16	5.26				
		2.02E-03	0.22	5.64E-13	4.55				

Ref .: Data reference.

Mode: Loading mode (controlled strain or controlled stress). K. A: Parameters of the general fatigue equation form  $(N=K^* - \Lambda)$ . K1, A1: Parameters of the fatigue equation form ( =K1\* N A1).

KSLA: Koninklijle Shekk Laboratorium Amsterdam

CRR: Road Research Center of Belgium

	Table (1) Continued.									
Ref.	Mode	K1	A1	K	A					
CRR	C. Stress	1.51E-03	0.20	7.85E-15	5.00					
	NEW 1 STR	1.18E-03	0.20	2.29E-15	5.00					
	autos	1.28E-03	0.19	5.95E-16	5.26					
	201.021.011	2.29E-03	0.27	1.67E-10	3.70					
	22.02.90	1.65E-03	0.23	7.98E-13	4.35					
3-42	11 11 11 11 11 11 11 11 11 11 11 11 11	1.87E-03	0.25	1.22E-11	4.00					
	20.00.00	2.13E-03	0.22	7.18E-13	4.55					
		1.21E-03	0.22	5.49E-14	4.55					
		1.31E-03	0.19	6.73E-16	5.26					
	राज प्रायः नाव	1.47E-03	0.20	6.86E-15	5.00					
	100.000.000	1.30E-03	0.19	6.46E-16	5.26					
	00 00 M	1.69E-03	0.21	6.30E-14	4.76					
	where an	2.36E-03	0.21	3.09E-13	4.76					
	25.14.55	2.28E-03	0.24	9.80E-12	4.17					
CRR	C. Strain	1.35E-03	0.20	4.48E-15	5.00					
		1.06E-03	0.19	2.21E-16	5.26					
	100 100 - 100	2.00E-03	0.24	5.68E-12	4.17					
	=	7.201:-04	0.15	1.12E-21	6.67					
	121.22	1.201-03	0.20	2.49E-15	5.00					
	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	9.80E-04	0.19	1.46E-16	5.26					
		1.19E-03	0.19	4.06E-16	5.26					
		1.10E-03	0.19	2.68E-16	5.26					
		1.63E-03	0.21	5.31E-14	4.76					
		1.02E-03	0.21	5.69E-15	4.76					
	at an a	1.56E-03	0.21	4.30E-14	4.76					
	2012c1 201	1.14E-03	0.19	3.24E-16	5.26					
	and the set	1.94E-03	0.23	1.61E-12	4.35					
	55 MS 707	1.38E-03	0.22	9.99E-14	4.55					
		1.36E-03	0.20	4.65E-15	5.00					
		1.63E-03	0.22	2.13E-13	4.55					
	Dis Dis TR	1.84E-03	0.22	3.69E-13	4.55					
	2228 89	1.82E-03	0.19	3.80E-15	5.26					
LCPC	C. Strain	1.75E-03	0.19	3.09E-15	5.26					
		1.17E-03	0.16	4.74E-19	6.25					
		1.30E-03	0.19	6.46E-16	5.26					
	Among Carr	7.40E-04	0.13	8.26E-25	7.69					
	A Statistic	8.50E-04.	0.15	3.38E-21	6.67					
		1.57E-03	0.20	9.54E-15	5.00					
	tarter	8.20E-04	0.16	5.14E-20	6.25					
1142	Philippi da	8.40E-04	0.15	3.13E-21	6.67					
		8.70E-04	0.16	7.45E-20	6.25					
		1.21E-03	0.20	2.59E-15	5.00					
- 11.9	2012/01/15	1.19E-03	0.14	1.29E-21	7.14					

# Table (1) Continued.

.

Ref.: Data reference.

Mode: Loading mode (controlled strain or controlled stress).

K, A: Parameters of the general fatigue equation form  $(N=K^*\square^A)$ .

K1. A1: Parameters of the fatigue equation form  $(\Box = K1*N^{-A1})$ .

LCPC: Laboratoire Central des Ponts et Chaussees-France

	1	Table (1) Co	ontinued.		
Ref.	Mode	K1	Al	К	A
LCPC	C. Strain	1.77E-03	0.19	3.28E-15	5.26
		1.09E-03	0.19	2.56E-16	5.26
		1.41E-03	0.16	1.52E-18	6.25
	an riska	1.78E-03	0.19	3.38E-15	5.26
	as to re-	4.29E-03	0.27	1.70E-09	3.70
	4.311	1.66E-03	0.22	2.31E-13	4.55
	in all the	2.81E-03	0.24	2.34E-11	4.17
		2.97E-03	0.25	7.78E-11	4.00
	===	1.02E-03	0.19	1.80E-16	5.26
Bonnaure	C. Strain			1.10E-06	3.10
et al. 1982	10/12/14			4.24E-07	3.31
	10000 278			1.36E-08	3.82
				2.48E-05	2.86
				5.40E-05	2.78
				1.94E-09	3.77
				1.82E-09	3.85
	17179 75			1.89E-07	3.31
100	IC. LT CO			1.62E-13	5.07
	Telefter			1.86E-13	5.11
	surrora .			6.76E-03	2.18
				1.62E-12	5.11
	-			1.87E-03	2.27
				5.56E-12	4.56
	22.7538			5.02E-11	4.30
	April 100 York			9.17E-13	4.83
	200 200 200			1.13E-13	5.14
	C. Stress			2.77E-11	4.47
				7.11E-11	4.34
Tayabafli					
et al. 1995	C. Strain			8.96E-08	3.57
Hsu et al.	C. Stress			6.11E-10	4.40
1996	20.00109			8.84E-11	5.50
1000	10.001			1.35E-12	5.50
1000				3.67E-13	5.50
15				1.55E-12	6.20
1	10.0013			2.94E-06	3.50
				5.39E-07	3.50
				1.69E-07	3.50
				5.61E-07	4.10
	.5. 2-1			4.30E-08	4,10
	11. marter			2.21E-08	4.10
-				1.08E-08	4.80
				7.40E-11	4.80
	===			3.76E-11	4.80

Data reference . Ref.

: Loading mode (controlled strain or controlled stress). Mode

: Parameters of the general fatigue equation form  $(N=K^*\square^{-A})$ . K, A

K1, A1: Parameters of the fatigue equation form  $(\Box = K1* N^{-A1})$ .

### FATIGUE EQUATION PARAMETERS RELATIONSHIPS

D.C.	Made	Table (1) C	No. 2 and Conference on Conference on	L K	A
Ref.	Mode	K1	Al	K LOOF OF	A 2.10
Hsu et al.	C. Stress			1.09E-05	3.10
1996				7.31E-07	3.80
	12 22.12			1.17E-07	3.80
				1.76E-08	3.80
				5.11E-08	4.20
Read	C. Stress			1.11E-09	3.79
ct al. 1997				3.89E-11	4.03
	201.122.202			7.18E-11	3.92
				1.65E-10	4.15
	12.02.72	1.36E-03	0.24	1.15E-12	4.17
	101 113 102	1.53E-03	0.25	5.45E-12	4.00
		1.14E-03	0.24	5.54E-13	4.17
	====	5.52E-04	0.18	7.94E-19	5.56
		5.09E-04	0.17	4.24E-20	5.88
		5.52E-04	0.17	6.84E-20	5.88
		6.24E-03	0.31	8.12E-08	3.22
	and viewants	6.01E-03	0.31	5.83E-08	3.26
		4.06E-03	0.32	3.18E-08	3.13
	TER BARNE	3.78E-03	0.31	1.63E-08	3.22
		1.60E-03	0.26	1.77E-11	3.85
	100 100 100	1.39E-03	0.22	1.02E-13	4.55
		1.50E-03	0.23	5.32E-13	4.35
	malers	3.52E-03	0.36	1.53E-07	2.78
		1.93E-03	0.27	8.79E-11	3.70
	871 YEAR	1.331-03	0.24	1.04E-12	4.17
Rowe	C. Stress			8.63E-07	2.64
et al. 1997	TOTAL			3.13E-06	2.65
				6.57E-05	2.23
				2.85E-07	2.75
112				2.51E-07	3.16
				2.18E-07	3.38
1000				1.58E-07	3.39
100	NI 1011			3.06E-09	3.78
				3.33E-10	
	and				4.30
	in the second			3.36E-09	4.04
				5.98E-10	3.89
				1.27E-09	3.67
				1.10E-11	4.27
Mar Dill				1.20E-07	3.12
Van Dijk	C. Strain			3.25E-14	5.38
et al. 1977	en artes		*****	1.17E-13	5.16
4	51.57 #1			6.12E-14	5.50
	200 ( 4000)			3.30E-19	7.37

### Table (1) Continued.

Ref.: Data reference.

Mode: Loading mode (controlled strain or controlled stress).

K, A: Parameters of the general fatigue equation form  $(N=K^*\square^{-\Lambda})$ .

K1, A1: Parameters of the fatigue equation form ( $\Box = K1 * N^{-A1}$ ).

(E

Table (1) Continued.									
Ref.	Mode	K1	AI	K	A				
Van Dijk	C. Strain			1.63E-13	4.93				
et al. 1977				6.03E-17	6.02				
	100 No. 111			4.27E-12	5.26				
	25 MG 85			2.63E-15	5.41				
	====			1.84E-18	6.12				
1000				1.59E-09	4.14				
	10130.0c			2.64E-06	2.96				
1001 1 2 2 2	2010/05			1.56E-08	3.97				
	707-02 215	*****		3.45E-14	5.07				
	10 10 to			9.52E-11	4.07				
	#2392.000			3.65E-22	7.59				
	urfa.			1.30E-16	6.34				
				3.86E-09	3.74				
	THE SAME IN-			6.36E-13	4.93				
	186 - 7 13 Tan			1.35E-12	5.03				
				4.15E-07	3.80				
• •	00-10.14			1.75E-17	6.07				
				2.65E-12	4.55				
				3.83E-10	4.38				
and the second second	====			1.03E-15	5.52				
	10 M M			1.35E-14	5.38				
1	100 pag 100			5.77E-13	5.07				
				1.45E-12	5.13				
				3.05E-12	4.57				
				1.46E-17	5.92				
				9.70E-05	2.79				
				2.10E-14	5.38				
				5.02E-09	4.11				
				2.64E-01	2.02				
	====			8.05E-13	4.94				
	mount an			9.20E-12	4.62				
Irwin 1977	C. Stress			8.56E-07	2.80				
				3.11E-06	2.90				
	- 224			5.60E-07	3.00				
'Yuce et al.	C. Stress			6.61E-07	3.16				
1974	===			2.19E-07	3.23				
	====			2.70E-07	2.96				
				1.72E-05	2.62				
				9.33E-07	3.08				
	were a			8.73E-08	3.75				
	2.23 M	*****		1.20E-08	3.75				
	12 ter ps			1.35E-06	3.05				
	====			3.79E-06	2.89				
	tor up on			6.32E-08	3.40				
and the second second second	And in case of the local division of the loc		No. of Concession, Name	0.020.00	5.40				

Table (1) Continued.

Ref.: Data reference.

Mode: Loading mode' (controlled strain or controlled stress).

K, A: Parameters of the general fatigue equation form (N=K\*13-A).

K1, A1: Parameters of the fatigue equation form ( =K1\* N<sup>-A1</sup>).

# Table (2) The Kolmogorov-Smirnov (K-S) test results

# 1- Variable: Parameter (A). D<sub>max</sub>= 0.57 Samples: (Kallas et al. 1971) and (Pell et al. 1975)

	SAMPLE [1]: (Kallas et al. 1971)			SAMPLE [2	SAMPLE [2]: (Pell et al. 1975)			
	Observed Frequency	Observed Relative Frequency	Observed Cumulative Frequency	Observed Frequency	Observed Relative Frequency	Observed Cumulative Frequency	D	
2.5-2.99*	3	.3	.3	5	.121	.121	.178	
3	5	.5	.8	7	.17	.292	.507	
3.5-3.99	1	.1	.9	7	.17	.463	.436	
4-4.49	0	0	.9	5	.121	.585	.314	
4.5-4.99	1	.1	1	8	.195	.78	.219	
5 -5.49	0	0.	1	4	.097	.878	.121	
5.55.99	0	0	1	4	097 .	.975	.024	
6 -6.49	0	0	1	1	.024	1	0	

2-

Variable: Parameter (A). D<sub>max</sub>=0.56 Samples: (Pell et al. 1975) and (CRGC)

*	SAMPLE [1	]: (Pell et al. 1	975)	SAMPLE [2	SAMPLE [2]: CRGC			
CI	Observed Frequency	Observed Relative Frequency	Observed Cumulative Frequency	Observed Frequency	Observed Relative Frequency	Observed Cumulative Frequency	D	
2.5-2.99	5	.121	.121	0	0	0	.121	
3	7	.17	.292	2	.048	.048	.243	
3.5-3.99	7	.17	.463	1	.024	.073	.39	
44.49	5	.121	.585	6	· · .146 ·	.219	.365	
4.5-4.99	8	.195	.78	I	.024	.243	.536	
55,49	4	.097	.878 ·	6	.146	.39	.487	
5.5-5.99	4	.097	.975	6	.146	.536	.439	
66.49	1	.024	1	2	.048	.585	.414	
6.5-6.99	0	0	1	1	.024	.609	.39	
7-7.49	0 .:	0	1 .	5	.121	.731	.268	
7.5-7.99	0	0	1	3	.073	.804	.195	
8-8.49	0	0	1	4	.097	.902	.097	
8.5-8.99	0	0	1	0	0	.902	.097	
99.49	0	0	1	2	.048	.951	.048	
9.5-9.99	0	0.	1	0	0	.951	.048	
10-10.5	0	0	1	2	.049	1	0	

3-

# Number 3 Volume 11 September 2005

Journal of Engineering

Variable

Variable: Parameter (A). Samples: (CRGC) and (KLSA)

D<sub>max</sub>=0.41

	SAMPLE [1	]: CRGC		SAMPLE [2	]: KLSA	• · · · ·	
CI	Observed Frequency	Observed Relative	Observed Cumulative	Observed	Observed Relative	Observed	D
	requency	Frequency	Frequency	Frequency	Frequency	Cúmulative Frequency	
2.5 - 2.99	0	0	()	1	.038	.038	.038
33.49	. 2	.048	.048	0	0	.038	.01
3.5-3.99	1	.024	.073	1	.038	.076	.003
4-4.49	6	.146	.219	3	.115	.192	.027
4.5-4.9%	1	.024	.243	2	.076	.269	.025
55.49	6	.146	.39	11	.423	.692	.302
5.5-5.99	6	.146	.536	5	.192	.884	.348
66.49	2	.048	.585	2	.076	.961	.376
6.5-6.99	1	.024	.609	0	0	.961	.351
7-7.49	5	.121	.731	1	.038	1	.268
7.5 - 7.99	3	.073	.804	0	0	1	.195
8-8.49	4	.097	.902	0	0	1	.097
8.5-8.99	0	0	.902	0	0	1	.097
99,49	2	.048	.951 -	()	0		.048
9.5-9.99	0	0	.951	0	0	1	.048
10—10.5	2	.049	1	0	0	1	0

4-

Variable: Parameter (A). Samples: (KLSA) and (CRR)

D<sub>max</sub>=0.42

	SAMPLE [1]	SAMPLE [1]: KLSA			SAMPLE [2]: CRR		
CI	Observed Frequency	Observed Relative Frequency	Observed Cumulative Frequency	Observed Frequency	Observed Relative Frequency	Observed Cumulative Frequency	D
2.5-2.99	1	.038	.038	0	0 .	0	.038
33,49	0	0	.038	0	0	0	.038
3.5-3.99	1	.038	.076 .	1	.027	.027	.049
44.49	. 3	.115	.192	5	.138	.166	.025
4.5-4.99	2	.076	.269	12	.333	.5	.23
5-5.49 •	· 11	.423	.692	17	.472	.972	.279
5.5-5.99	5 :	.192	.884	0	0	.972	.087
66.49	2	.076	.961	0	0	.972	.01
6.5-6.99	0	0	.961	1	.027	1	.038
77.49	I	.038	1	0	0	1	0

FATIGUE EQUATION PARAMETERS RELATIONSHIPS

5-

## Variable: Parameter (A). Samples: (CRR) and (LCPC)

D<sub>max</sub>=0.45

	SAMPLE [1	]: CRR		SAMPLE [2			
CI	Observed Frequency	Observed Relative Frequency	Observed Cumulative Frequency	Observed Frequency	Observed Relative Frequency	Observed Cumulative Frequency	D
3.53.99	1	.027	.027	1	.05	.05	.023
44.49	5	.138	.166	3	.15	• .2	.034
4.5 - 4.99	12	.333	.5	2		.3	.2
55.49	17	.472	:172	6	5	.6	.372
5.5-5.99	• 0 •	0	.972	0	0	.6	.372
6-6.49	0	0	.972	4	.2	.8	.172
6.5-6.99	1	.027	1	2	.1	.9	.099
7—7.49	0	0	1	1	.05	.95	.049
7.57.99	0	0	1	1	.05	1	0

6-

Variable: Parameter (A). D<sub>max</sub>=0.52 Samples: (LCPC) and (Bonnaure et al. 1982)

	SAMPLE [1	]: LCPC,		SAMPLE [2]: Bonnaure et al. (1982)			
CI	Observed	Observed	Observed	Observed	Observed	Observed	D
	Frequency	Relative	Cumulative	Frequency	Relative	Cumulative	200
		Frequency	Frequency		Frequency	Frequency	
22.49	0	()	0 .	2	.105	.105	.105
2.5-2.99	0	0	0	2	.105	.21	.21
3-3.49	0	0	0	3	.157	.368	.368
3.5-3.99	1	.05	.05	3	.157	.526	.476
44.49	3	.15	.2	3	.157	.684	.484
4.5-4.99	2	.1	.3	2	.105	.789	.489
5-5.49	6	.3	.6	4	.21	1	.399
5.5-5.99	0	0	.6	0	0	1	.399
66.49	4	.2	.8	0	0	1	.199
6.5-6.99	2	.1,	.9	0	0	1	.099
77.49	1	.05	.95	0	0	. 1	.049
7.5-7.99	1	.05	1	0	0	1	0

Number 3 Volume 11 September 2005

Variable: Parameter (A). D<sub>max</sub>=0.53 Samples: (Bonnaure et al. 1982) and (Hsu et al. 1995)

	SAMPLE [1	]: Bonnaure et	t al. (1982)	SAMPLE [2	]: Hsu et al. (1	995)	
CI	Observed Frequency	Observed Relative Frequency	Observed Cumulative Frequency	Observed Frequency	Observed Relative Frequency	Observed Cumulative Frequency	D
2-2.49	2	.105	.105	0	0	0	.105
2.5-2.99	2	.105	.21	0	0	0	.21
3-3.49	3	.157	.368	1	.052	.052	.315
3.5-3.99	3	.157	.526	6	.315	.368	.157
44.49	3	.157	.684	5	.263	.631	.052
4.5-4.99	2	.105	.789	3	.157	• .789	0
5-5.49	4	.21	1	0	0	.789	.21
5.5-5.99	0	0	1	3	.157	.947	.052
66.49	· 0 ·	0	1	1	.052	.999	0

8-

7-

Variable: Parameter (A). D<sub>max</sub>=0.52 Samples: (Hsu et al. 1995) and (Read et al. 1997)

CI	SAMPLE [1]: Hsu et al. (1995)			SAMPLE [2]: Read et al. (1997)			
	Observed Frequency	Observed Relative Frequency	Observed Cumulative Frequency	Observed Frequency	Observed Relative Frequency	Observed Cumulative Frequency	D
2.5-2.99	0	0,	0	1	.05	.05	.05
3-3.49	1	.052	.052	4	.2	.25	.197
3.5-3.99	6	.315	.368	4	.2	.45	.081
4-4.49	5	.263	.631	7	.35	.799	.168
4.5-4.99	• 3 •	.157	.789	-1	.05	.849	.06
5-5.49	0	0	.789	0	0	.849	.06
5.5-5.99	3	.157	.947	3	.15	1	.052
6-6.49	1 :	.052	.999	0	0	1	0

9-

Variable: Parameter (A). D<sub>max</sub>=0.57 Samples: (Read et al. 1997) and (Rowe et al. 1997)

	SAMPLE [1]: Read et al. (1997)			SAMPLE [2			
CI	Observed Frequency	Observed Relative Frequency	Observed Consolative Frequency	Observed Frequency	Observed Relative Frequency	Observed Cumulative Frequency	D
2-2.49*	0	0	0	1	.071	.071	.071
2.5-2.99	1	.05	.05	3	.214	.285	.235
3-3.49	4	.2	.25	4	.285	.571	.321
3.5-3.99	4	.2	.45	3	.214	.785	.335
4-4.49	7	.35	.799	3	.214	1	.2
4.5-4.99	1	.05	.849	0	0	1	.15
5-5.49	0	0	.849	0	0	1	.15
5.5-5.99	3	.15	1	0	0	1	0

### 10-

Variable: Parameter (A). D<sub>max</sub>=0.52 Samples: (Rowe et al. 1997) and (Van Dijk et al. 1977)

	SAMPLE [	1]: Rowe et a	ul. (1997)	SAMPLE [	2]: Van Dijk	et al. (1977)	
CI	Observed	Observed	Observed	Observed	Observed	Observed	D
A REAL	Frequency	Relative	Cumulative	Frequency	Relative	Cumulative	
	:	Frequency	Frequency		Frequency	Frequency	
2-2.49	1	.071	.071	1	.028	.028	.042
2.5— 2.99	3	.214	.285	2	.057	.085	.2
3	4	.285	.571	0	0	.085	.485
3.5— 3.99	3	.214	.785	7	.2	.285	.5
44.49	3	.214	1	8	.228	• .514	.485
4.5— 4.99	0	0	1	4	.114	.628	.371
55.49	0	0	1 1	4	.114	.742	.257
5.5— 5.99	0	0	1	3	.085	.828	.171
6-6.49	0	0	1	4	.114	.942	.057
6.5— 6.99	0 :	0	1	0	0	.942	.057
7-7.49	0	0	1	1	.028	.971	.028
7.5— 7.99	0	0	1	1	.028	1	0

618

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# Number 3 Volume 11 September 2005

11-

Variable: Parameter (A). D<sub>max</sub>=0.58 Samples: (Van Dijk et al. 1977) and (Yuce et al. 1974)

	SAMPLE [	1]: Van Dijk	et al. (1977)	SAMPLE [	2]: Yuce et al	l. (1974)	
CI	Observed	Observed	Observed	Observed	Observed	Observed	D
	Frequency	Relative	Cumulative	Frequency	Relative	Cumulative	
		Frequency	Frequency		Frequency	Frequency	
2-2.49	1	.028	.028	0	0.	0	.028
2.5-	2	.057	.085	3	.3	.3	.215
3-3.49	· 0 ·	0	.085	3	.3	.6	.515
3.5— 3.99	7	.2	.285	1	.1	.7	.415
4-4.49	8	.228	.514	2	.2	.9	.386
4.5— 4.99	4	.114	.628	1	.1	1	.372
5-5.49	4	.114	.742	0	0	1	.257
5.5— 5.99	3	.085	.828	0	0	1	.171
66.49	4	.114	.942	0	0	1	.057
6.5— 6.99	0	0'	.942	0	0	e 1	.057
77.49	1	.028	.971	0	0	1	.028
7.5— . 7.99	1	.028	1 •	0	0	I	0